COSMIC Capstone Project Final Briefing

Zenith Engineering Solutions, University of Texas at Arlington: Autonomous In-Space Truss Assembly Robot (AITAR)

Students: Ethan Cross, Spencer Cornwell, Juan Rodriguez, Nicholas Hermes, Nevin Puthenpurackal Advisor: Dr. Paul Davidson, UTA Mentor: Dr. Jacob Rome, Aerospace Corp

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Mechanical and Aerospace Engineering

THE UNIVERSITY OF TEXAS AT ARLINGTON

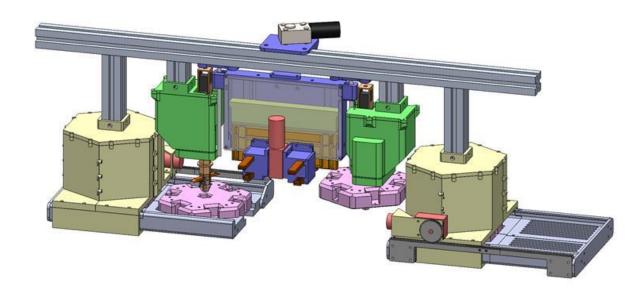




Executive Summary

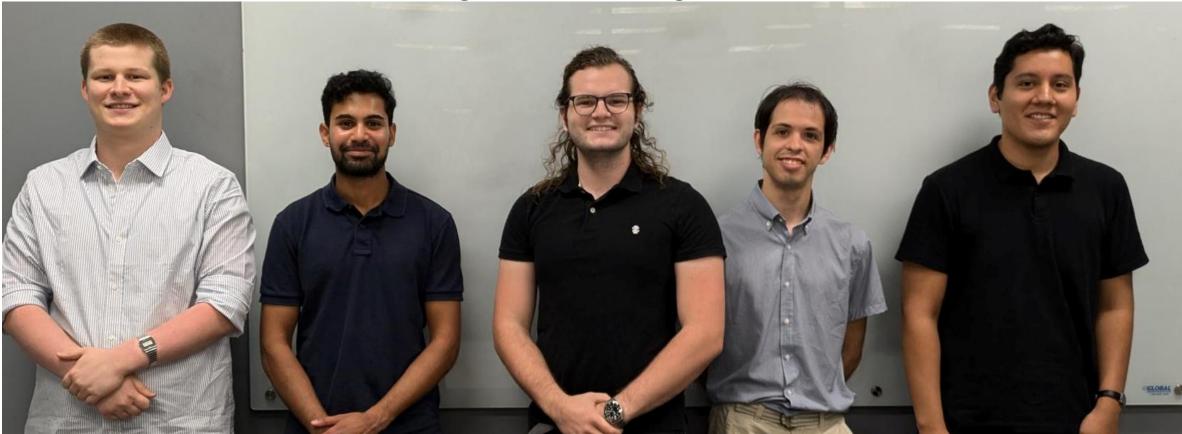
Autonomous In-Space Truss Assembly Robot (AITAR)

- Motivation
 - Limited operational life of space systems
 - Need larger payloads
 - Trusses assembled by humans
- Modular
- How does it solve the problem?
 - Enables larger and more capable spacecrafts
 - Automates truss assembly in space
- Prototype phase





Team Overview - Undergraduate Engineers



Spencer Cornwell Beam Magazine Beam & Node CAD & Manufacturing Nevin Puthenpurackal Beam & Node Documentation Ethan Cross Team Lead Programming Mechatronics Nicholas Hermes Node Magazine Animation Juan D. Rodriguez Torque Arm Node Magazine CAD & Manufacturing



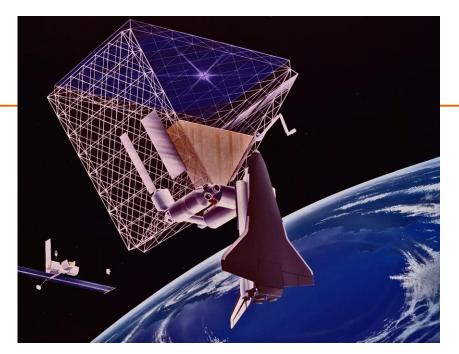
2.4 Systems Engineering Milestones

Milestone	Date	
Select program manager	10-10-2024	
Chose operations/capability	10-17-2024	
Defined system requirements	11-12-2024	
Completed trade studies	10-31-2024	
Finalized conceptual design	11-21-2024	
Developed a path to Preliminary Design review	04-05-2025	
Generated prototype specifications	02-05-2025	
Prototype sub-assemblies functional	04-03-2025	
Prototype assembled	04-07-2025	



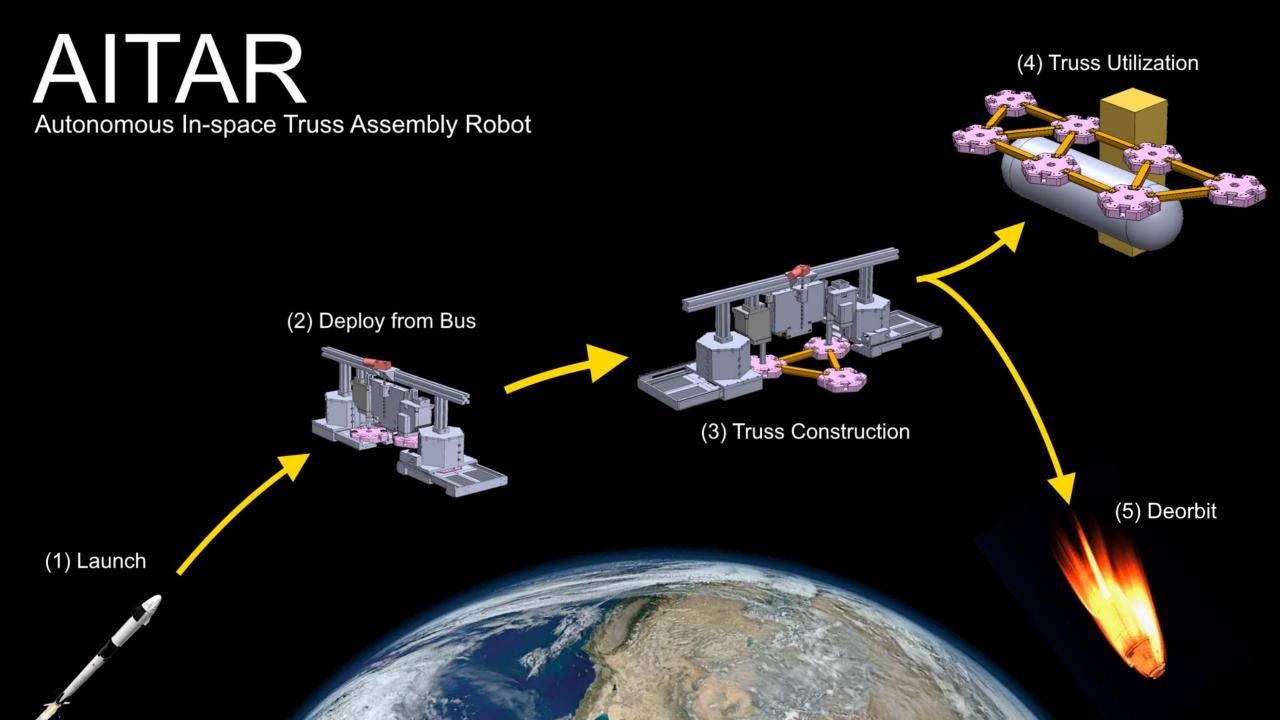
1.1 Impact

- Enables in-space structures larger than payload volume
- Serve as a support structure for repair and assembly
- Platform to enable technologies in space infrastructure
 - Structural
 - Fuel

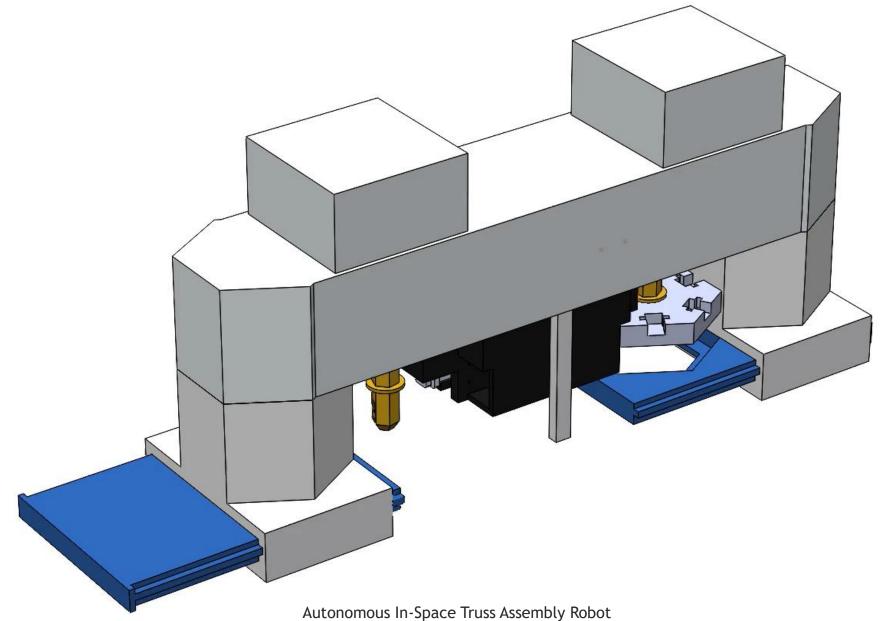




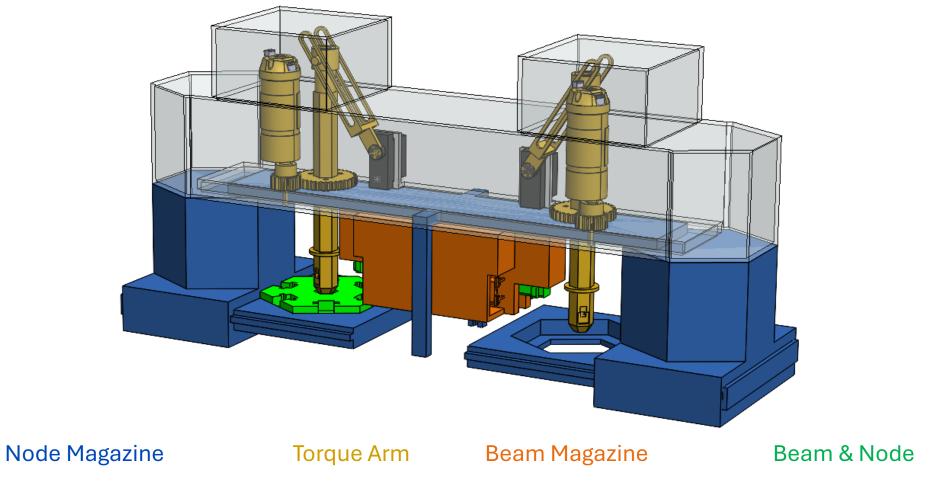




2.1 Animation of Key Operating Sequence



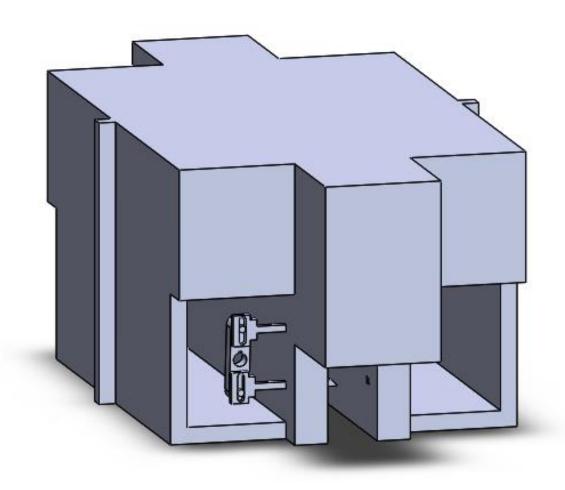
Subassemblies - Conceptual Design





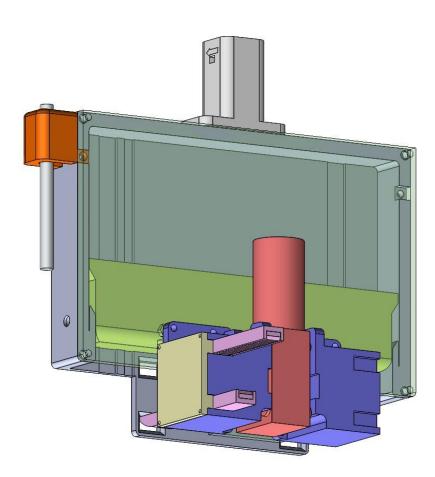
Beam Magazine - Conceptual

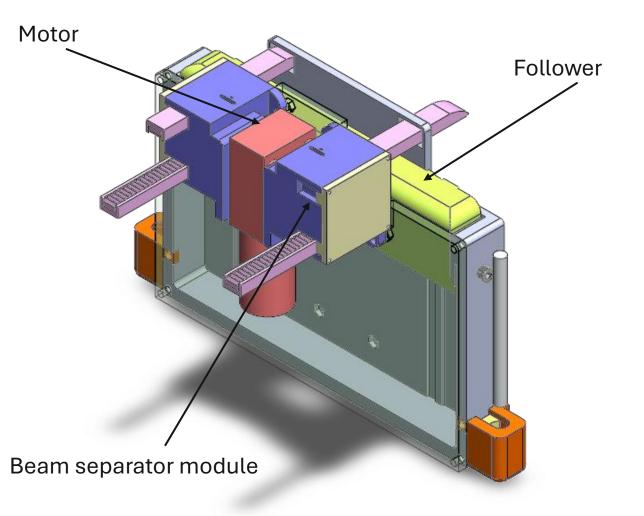
- Store
- Dispense
 - Spring and follower
- Isolate
 - Servomotor
- Translate
 - Ball Screw
 - DC motor



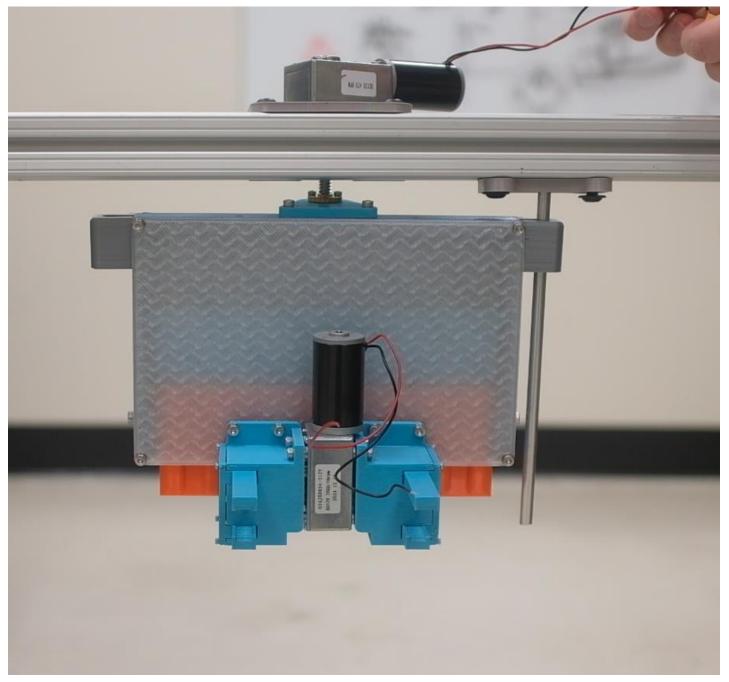


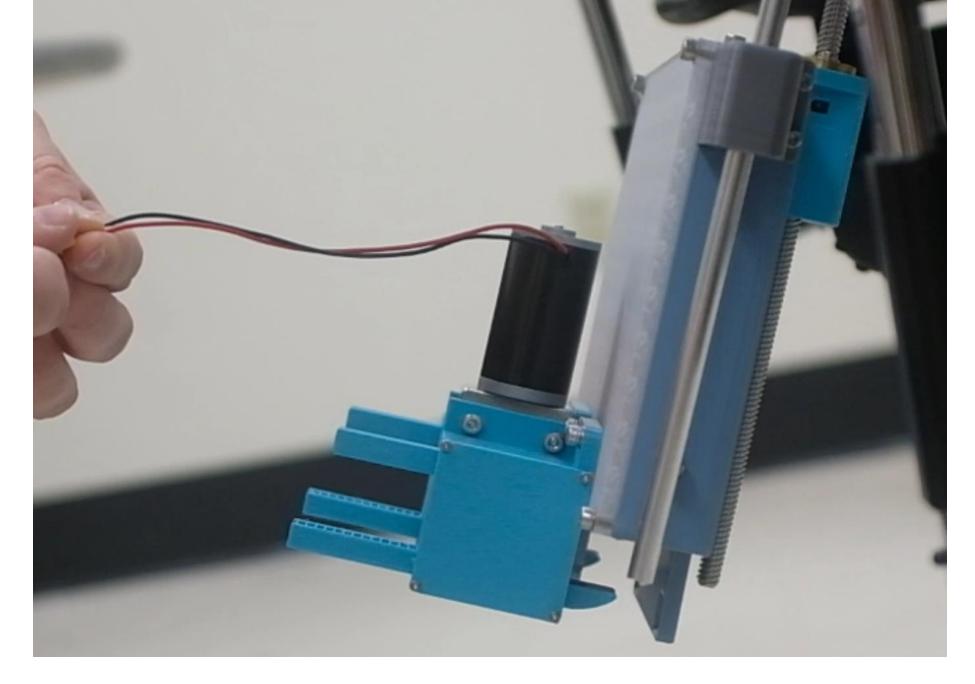
Beam Magazine - Prototype





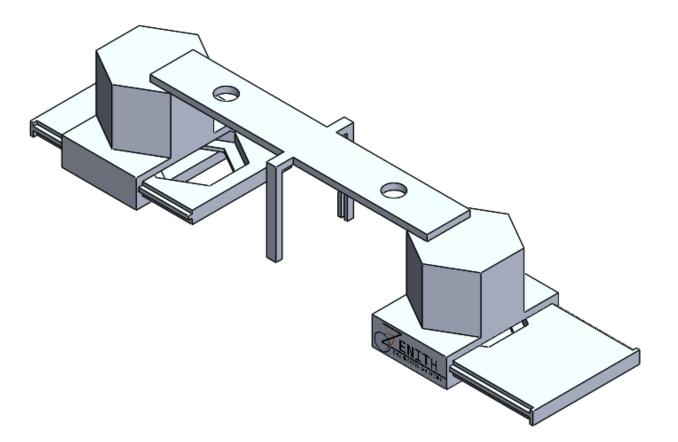






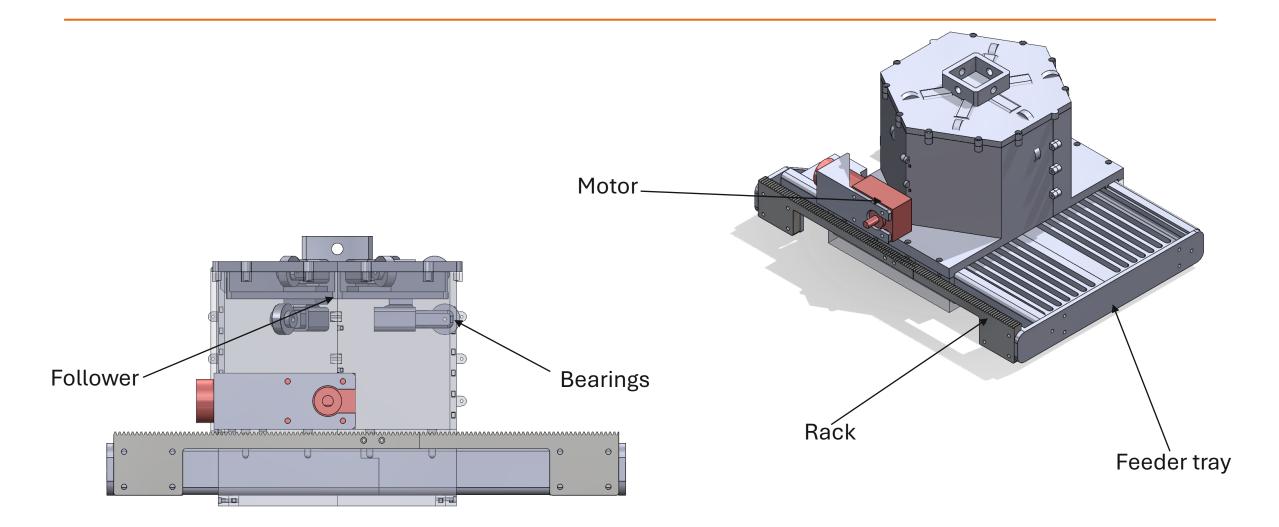
Node Magazine - Conceptual

- Store
- Dispense
 - Ball Screw
 - DC motor

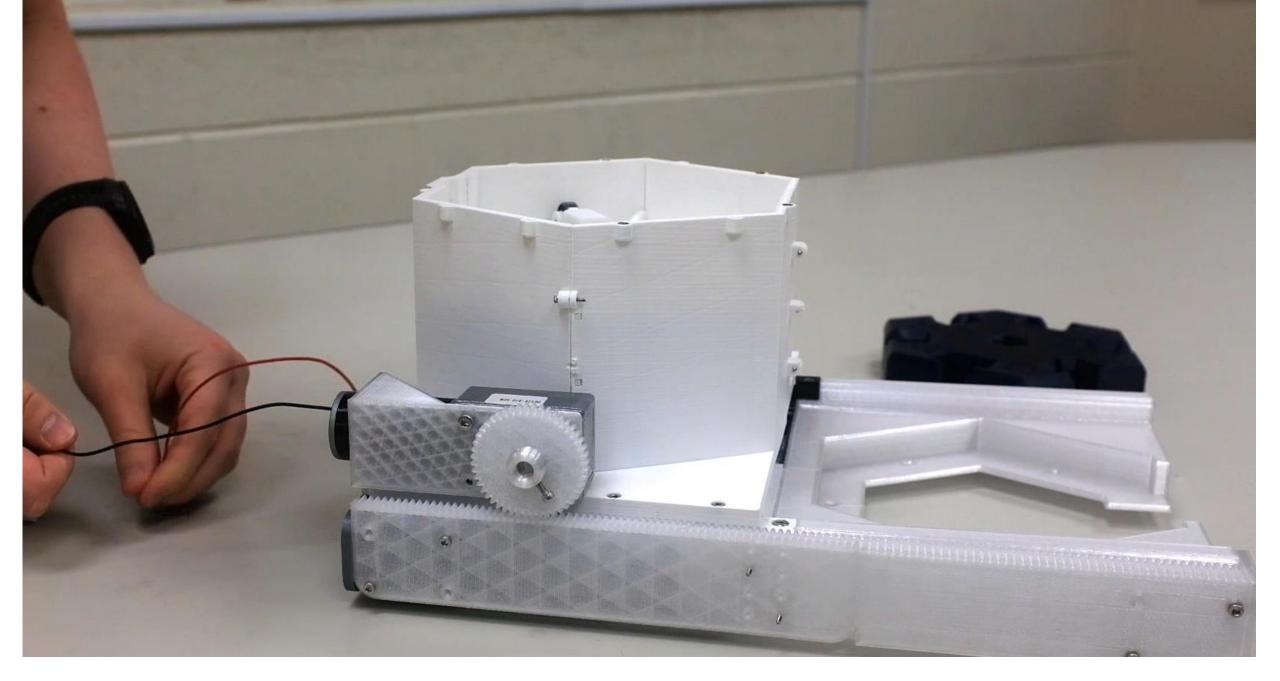




Node Magazine - Prototype

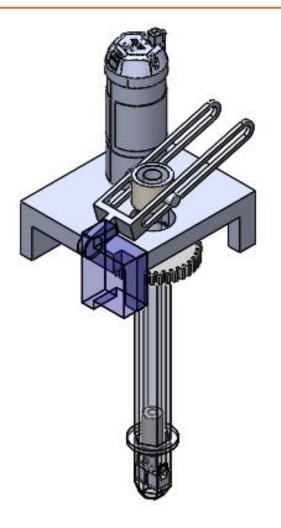






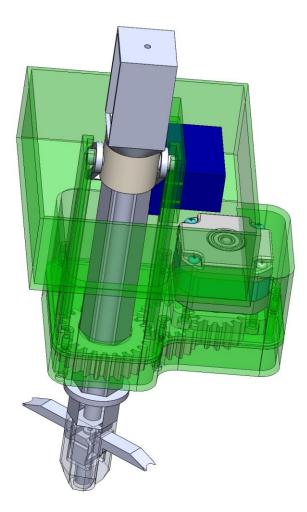
Torque Arm - Conceptual

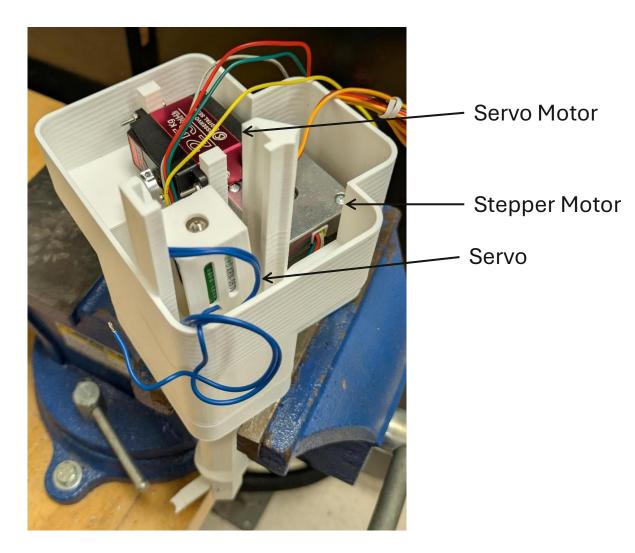
- Constrain
 - Flanges
 - Servo motor
- Translate
 - Servo motor
- Rotate
 - Stepper motor
 - Gears





Torque Arm - Prototype



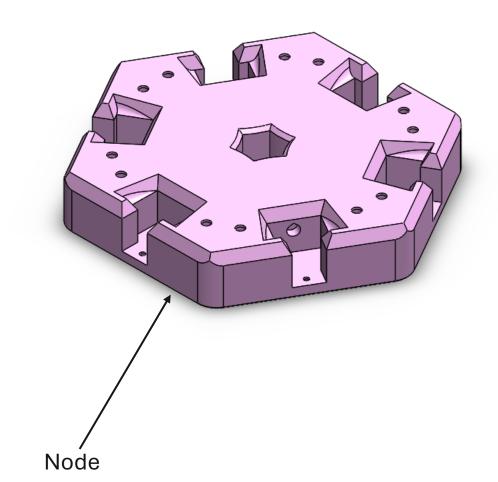






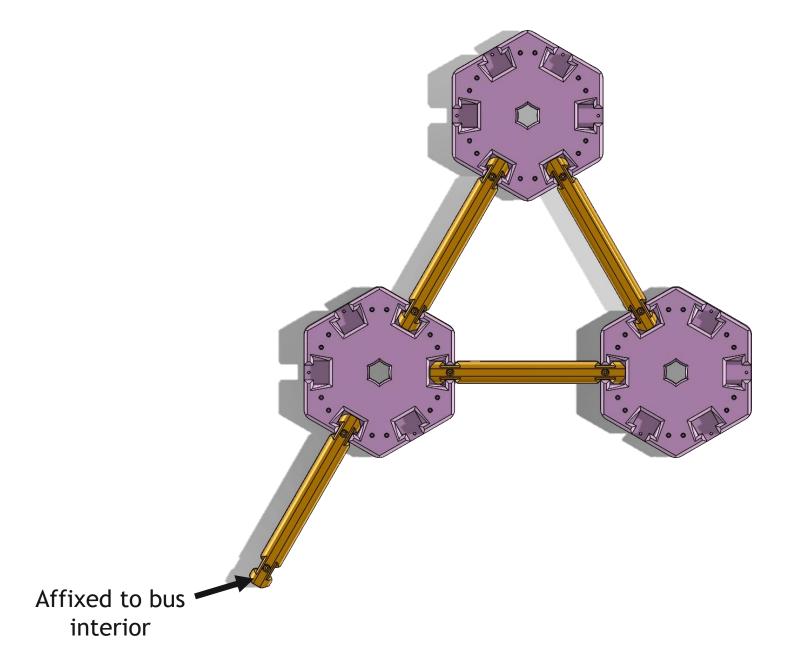
Truss Elements

- Induction welding
- PEEK/CF beam
 - Metal in ends



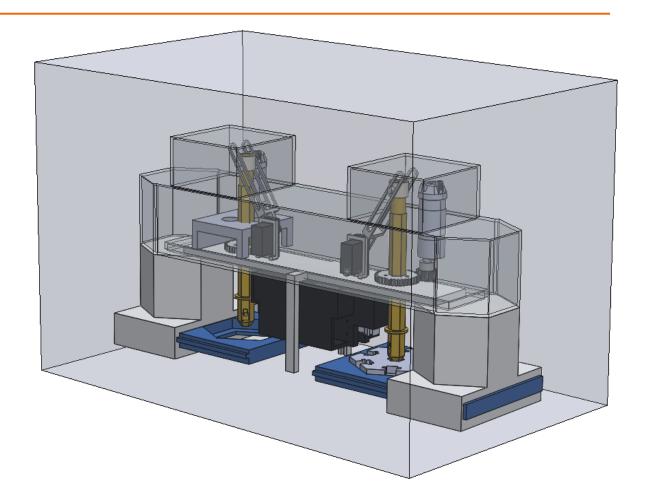


Beam



Completion of Required Elements

- Mass: 17 kg
- Power
 - >222W welding power draw
- Max Launch stress of 13 MPa at 6 G's
- Operations
 - Dispensing/Alignment
 - Joining
 - Locomotion





1.2 Feasibility

- Kinematics proven
- Automation
- BOM Critical components:
 - Stepper motors
 - Servo motors
 - Ball screws
 - Induction coil
 - Composite beams

Feasible within a decade



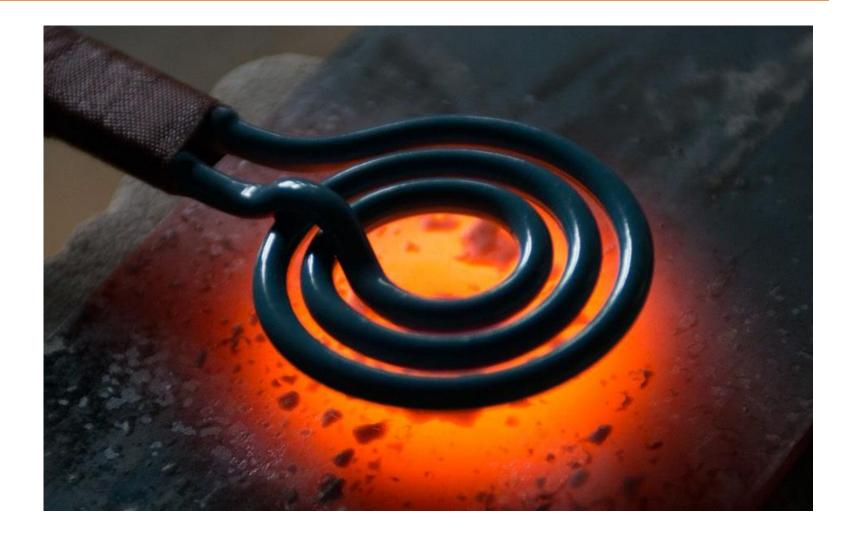
3.2 Technology Gap Assessment

- Thermoplastic induction welding not subject of recent research for space applications
- Capability to inspect welded joints would ensure the stability of the truss
- Streamlining reloading would decrease build time and failure rate
- No technology gap which is insurmountable



1.7 Trade Studies - Joining

- Fasteners
- Ultrasonic welding
- Adhesives
- Snap fits
- Induction welding





1.7 Trade Studies - Linear Actuation

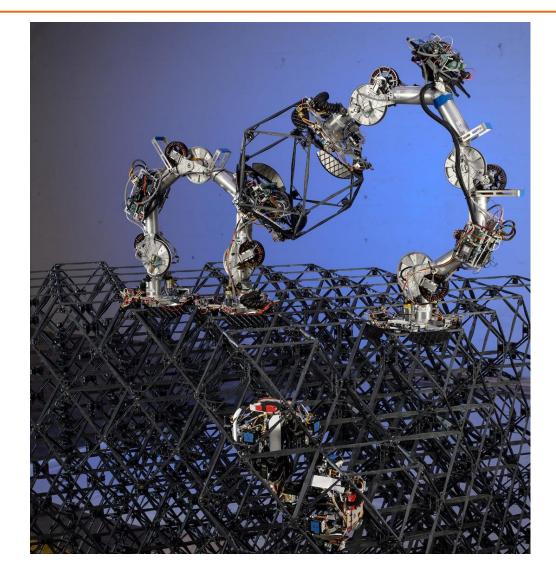
- Lead screw
 - Rely on sliding friction
- Ball screw
 - Reduced friction
 - Used in space applications





1.3 Innovation

- Similar projects under development
 - ARMADAS
 - NASA Assemblers
 - SAMURAI
- What makes ours unique?
 - Simple
 - Modular
 - On-orbit
 - Scalable
 - Plastic welding





1.5 Risk Assessment

		Severity					
Probability		4	3	2	1		
	4			Electronic interruptions			
	3	Misalignment of nodes and beams	Welded joint breaks	Robot loses communication			
	2	Follower motion impeded		Actuator failure	Insufficient power generation		
	1	Impact with space debris	Beam and node discharge		Regular wear and tear		



2.3 Data Handling and Comms

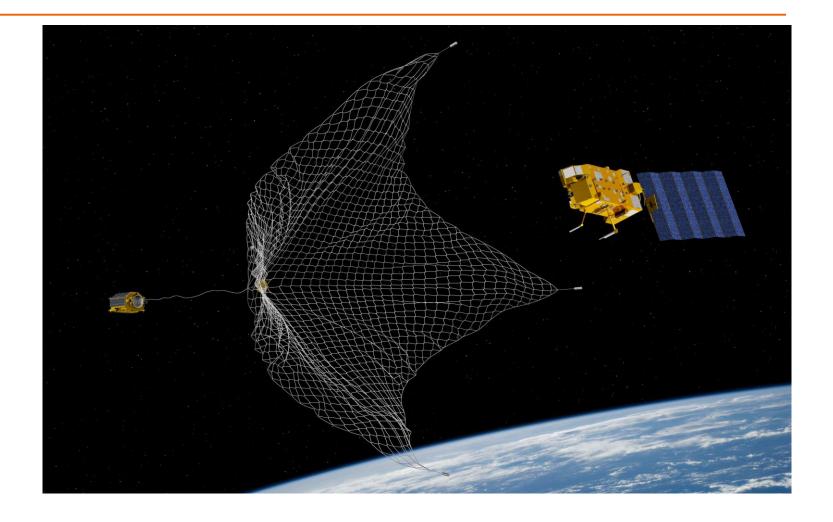
- No-need for real time
- Could be constantly observed, or piecewise
- Operator only needed for initialization and updates if desired
- Low bitrate
- **Brief** skimming of systems brought us to: Astrodev Lithium-1
 - 9.8 Kbps
 - 48g
 - 0.25-4W Tx Power





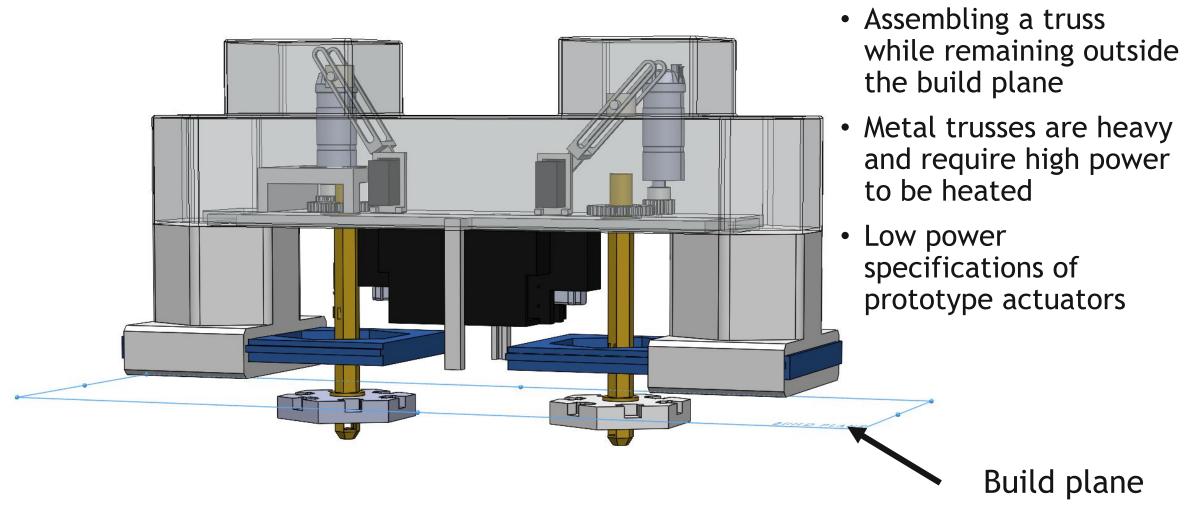
3.1 Most Innovative Concepts Considered

- Satellite commandeering
- Solar panel repair
- Induction welded composite trusses



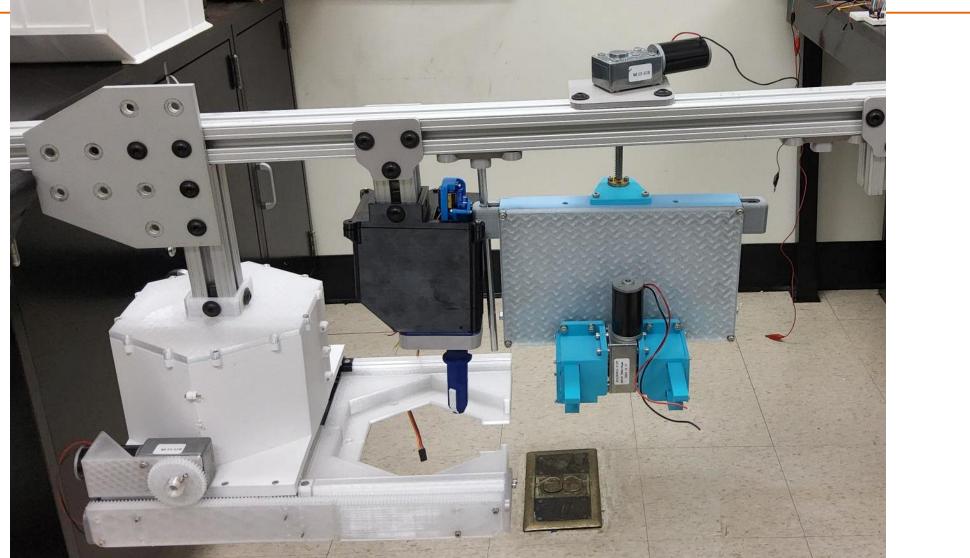


3.3 Biggest Challenges





Prototype





4.1 Paper

- Discusses Detailed decisions and motivations
- Abstract: 163 words
- Paper: 13 pages
- References: 26
- Can be published at: Journal of Spacecraft and Rockets



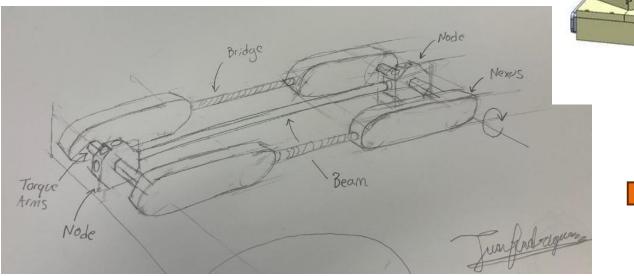
1.6 Path to PDR

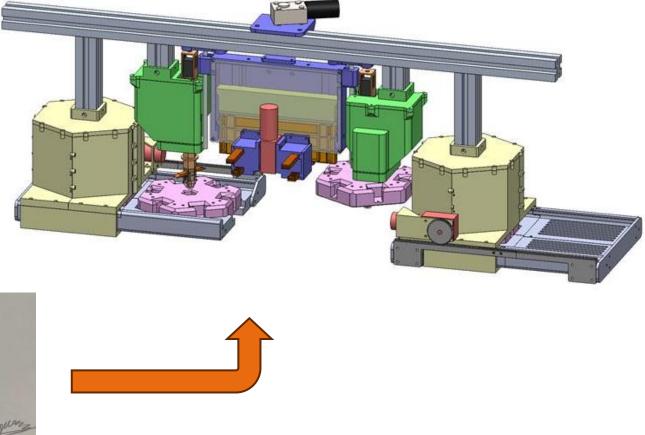
- Design launch configuration and mounting structure to bus
- Design disassembly and reloading processes
- True environment testing
 - Resistance to rotation resulting from the robot's locomotion
 - Control of thermal loading from welding and motion
- Rescale to fit bus
- Assignment of space-rated motors



Summary/Conclusions/Highlights

- Research
- Mission Selection
- Iterations
- Conceptual Design
- Prototype







Questions?

